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# Linear Solvers for Adjoint Problems

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Presenter:  
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MADELEINE project,  
Grant Agreement No 769025

# Outline

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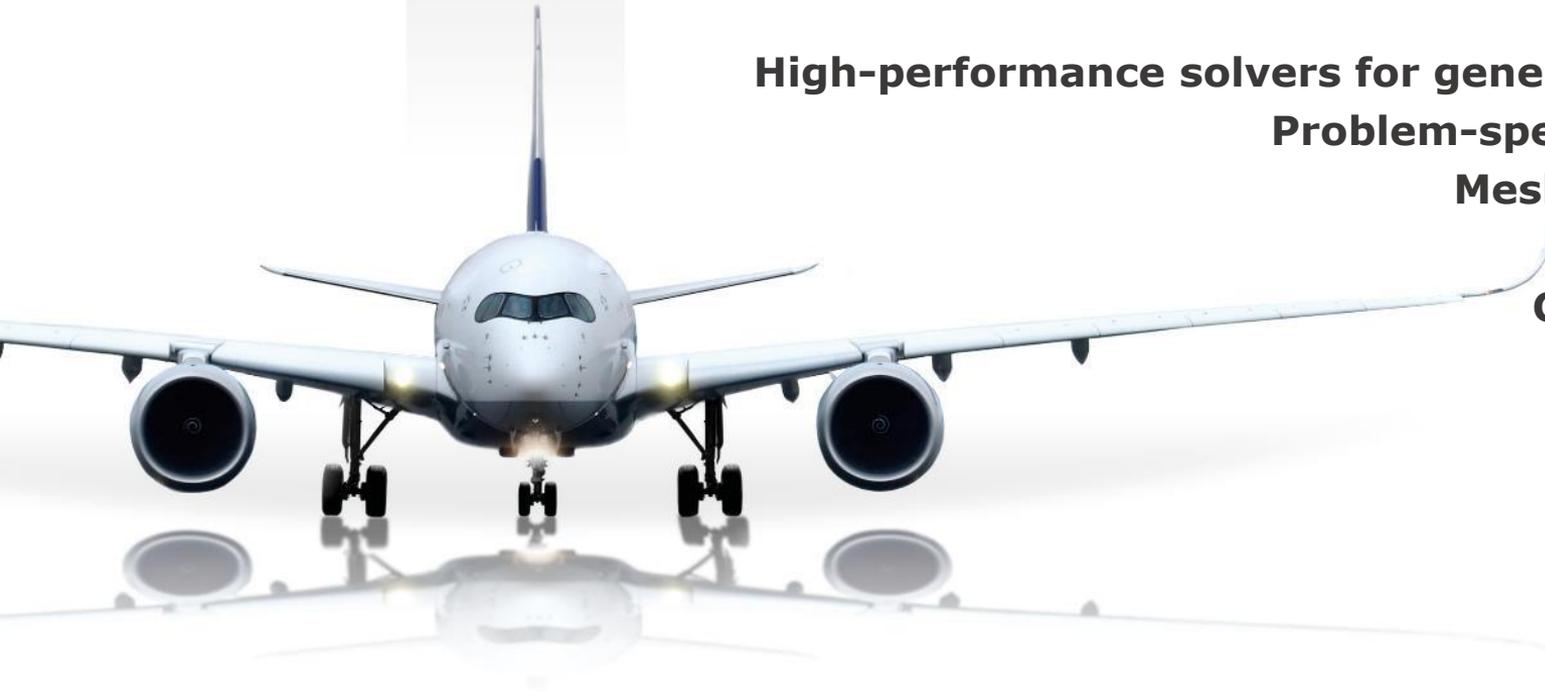
**Introduction**

**High-performance solvers for general adjoints**

**Problem-specific solver**

**Mesh morphing**

**Conclusions**



# The role of linear solvers

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# Motivations

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## Linear solvers are at the heart of adjoint based optimization

- used to solve the discrete adjoint equations
- used to propagate mesh deformation from boundary to entire computational domain

## From single to multi discipline (Madeleine)

- used to solve the system of systems of coupled adjoints (disciplinary adjoints + coupling)

# Challenges

- **cost ->  $10^8$  degrees of freedom;**
  - algorithmic complexity
  - memory requirements
  - scaling and distributed computation especially on next-gen HW

vs

- **robustness**
  - convergence in complex situations, problem dependency
  - usability, especially in multi-disciplinary setting
  - accuracy of resulting gradients

Maintainability

# Objectives

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## Efficient MDO process

- accelerate the **execution time** of the MDO process by a **factor of 5** by exploiting HPC capabilities, optimize code, develop alternative algorithms

## Maintainability

- build upon and integrate into (widely-used) existing framework
- to factorize efforts, especially towards next-gen HW and energy efficiency

# Outline

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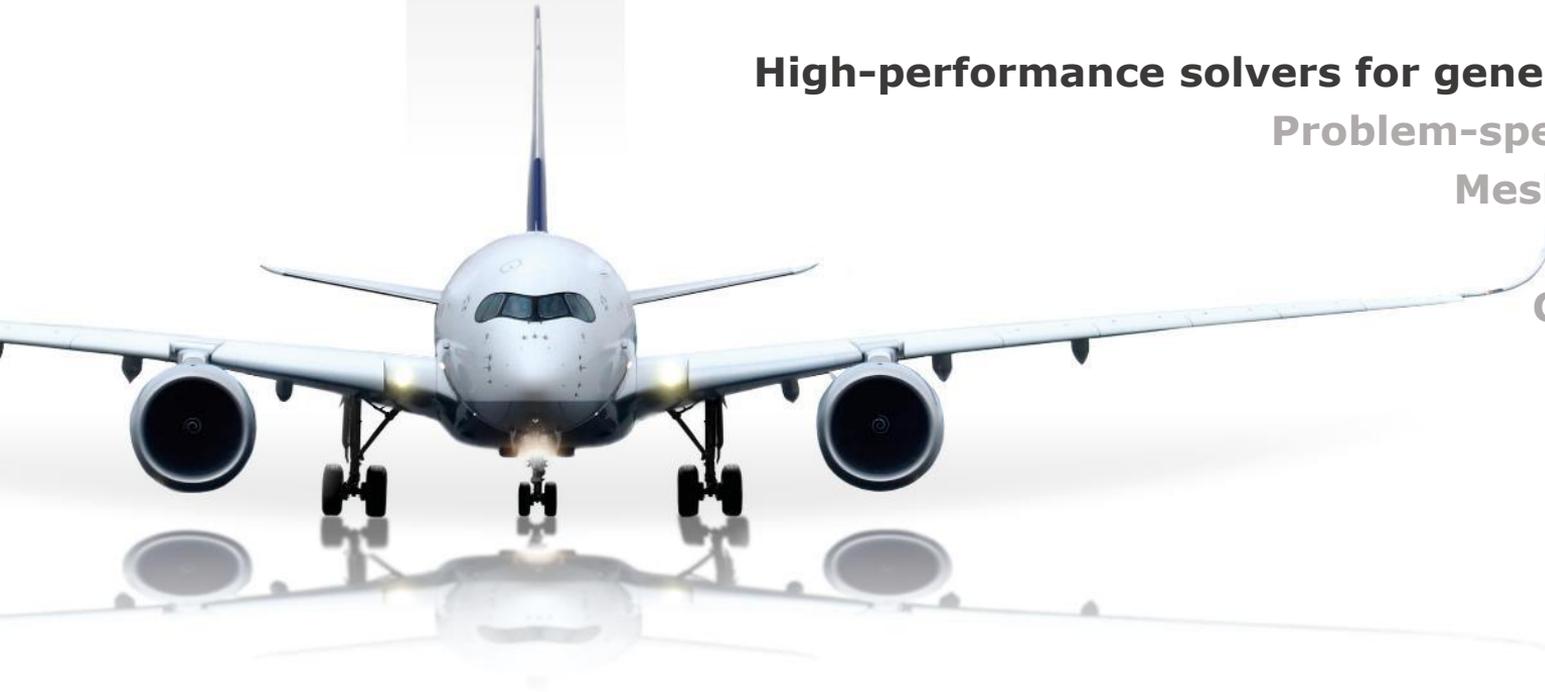
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Problem-specific solver

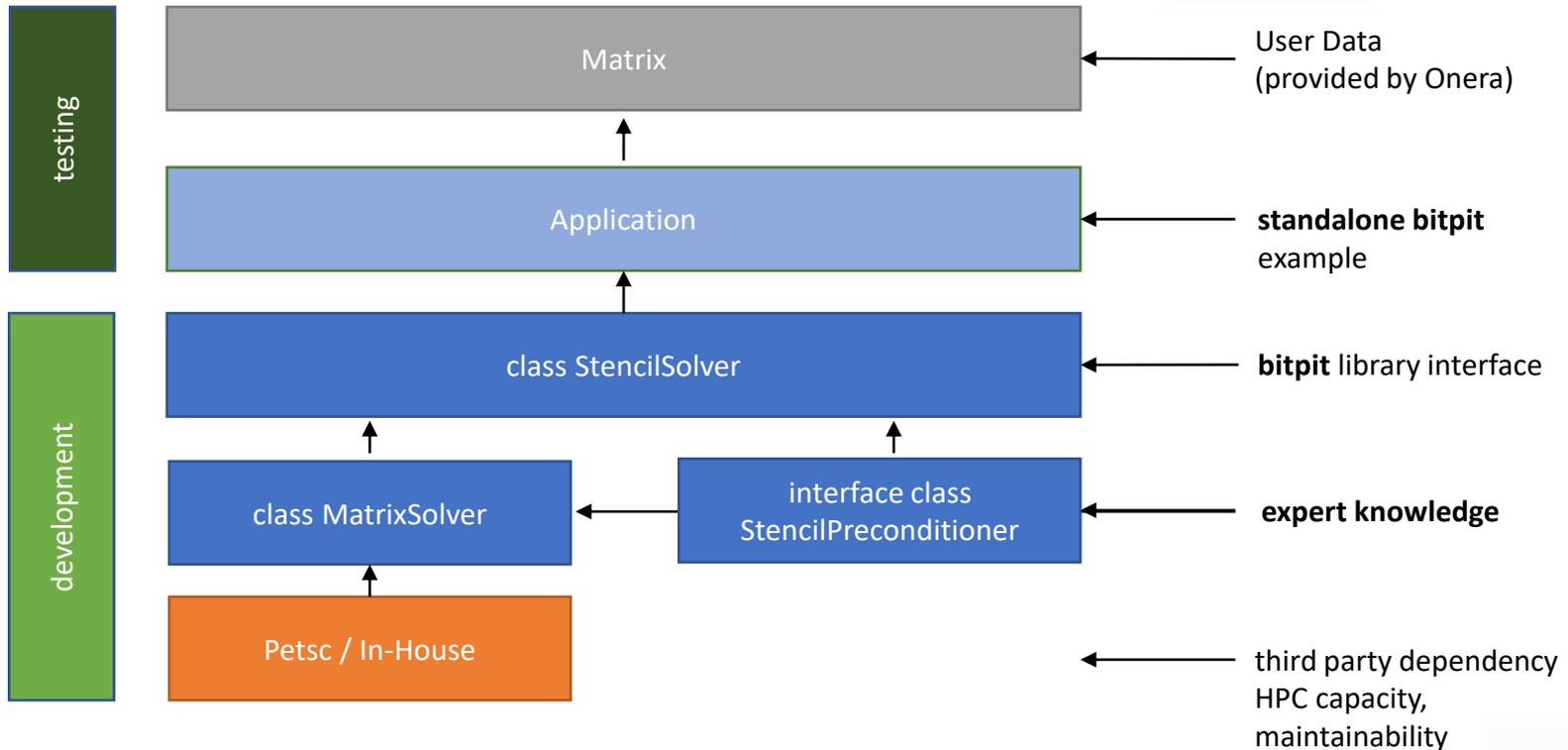
Mesh morphing

Conclusions



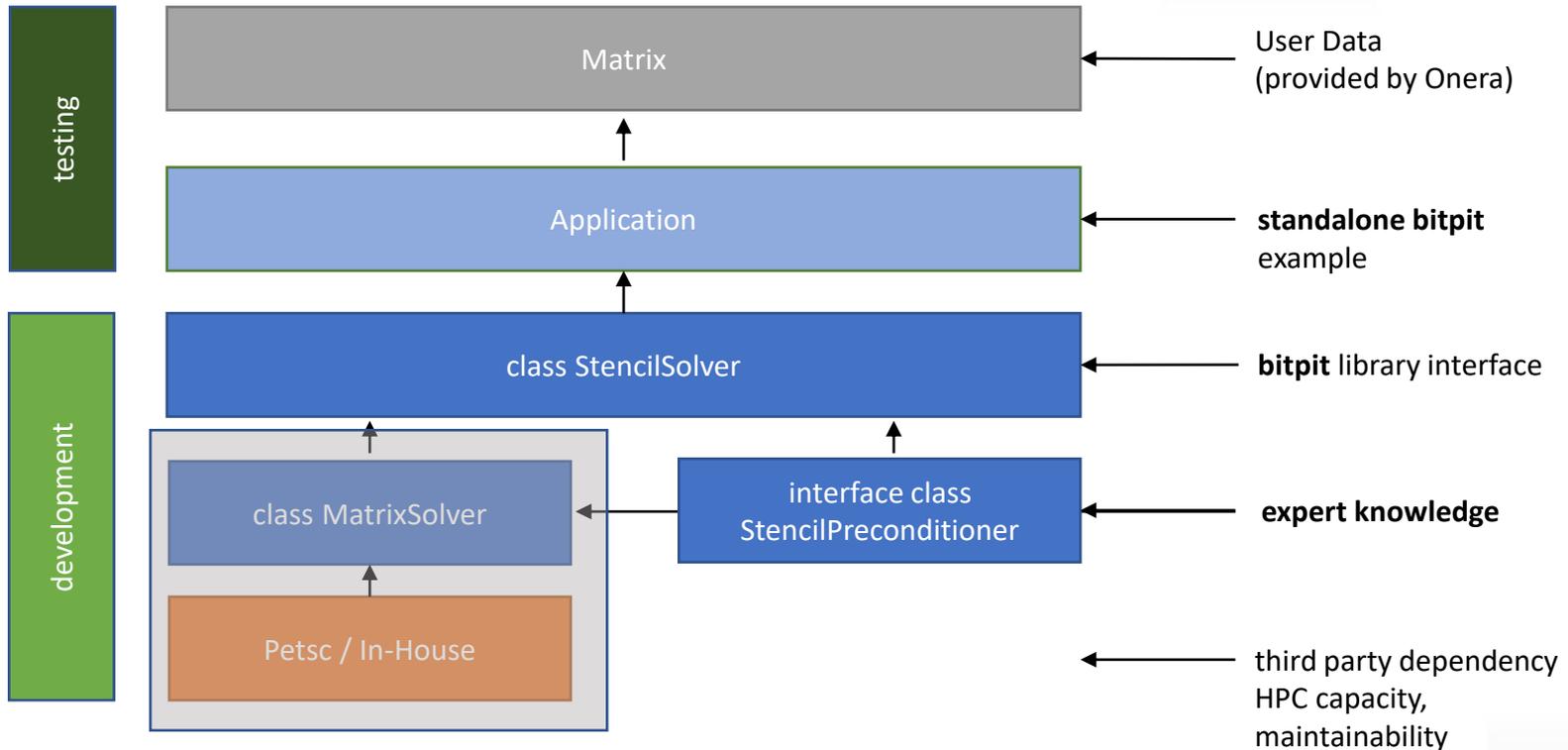
# Framework

- Different layers for iterative solvers, preconditioners and testing



# 1 Optimized iterative solvers

- Exploitation of Flexible GMRES and Deflation techniques



# Test case

2D compressible RANS equations

Transonic turbulent flow over OAT15A profile

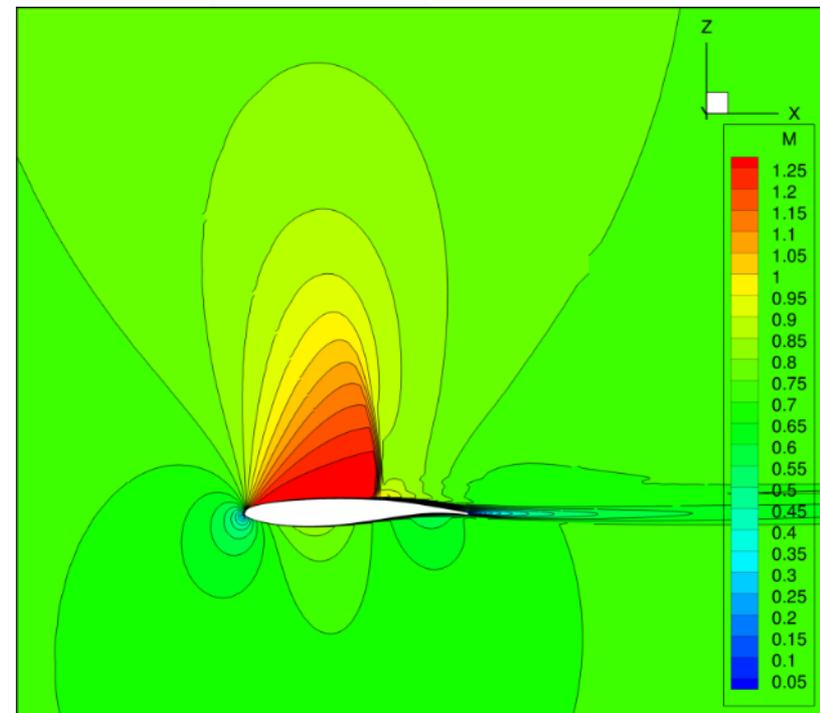
- $M=0.73$ ,  $\alpha=2.5^\circ$ ,  $Re=3 \times 10^6$

Space discretization

- Mesh is composed of 64416 hexahedra
- 4th order DG scheme

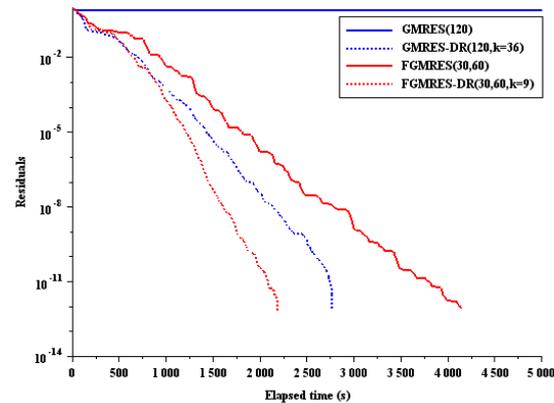
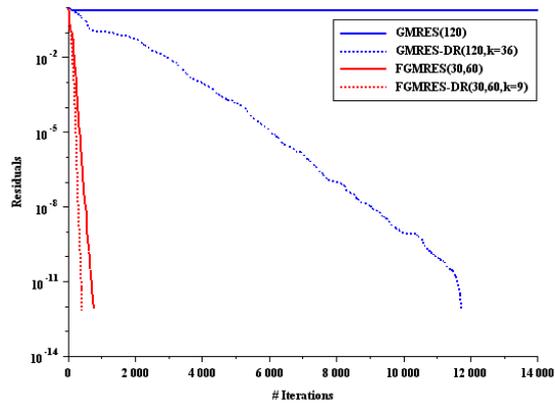
Aerodynamic shape optimization problem

- Minimization of drag coefficient



# 1 Enhanced iterative solvers, Flexible GMRES +deflation

- 4<sup>th</sup> order DG scheme with 12 MPI processes
- memory foot-print kept constant

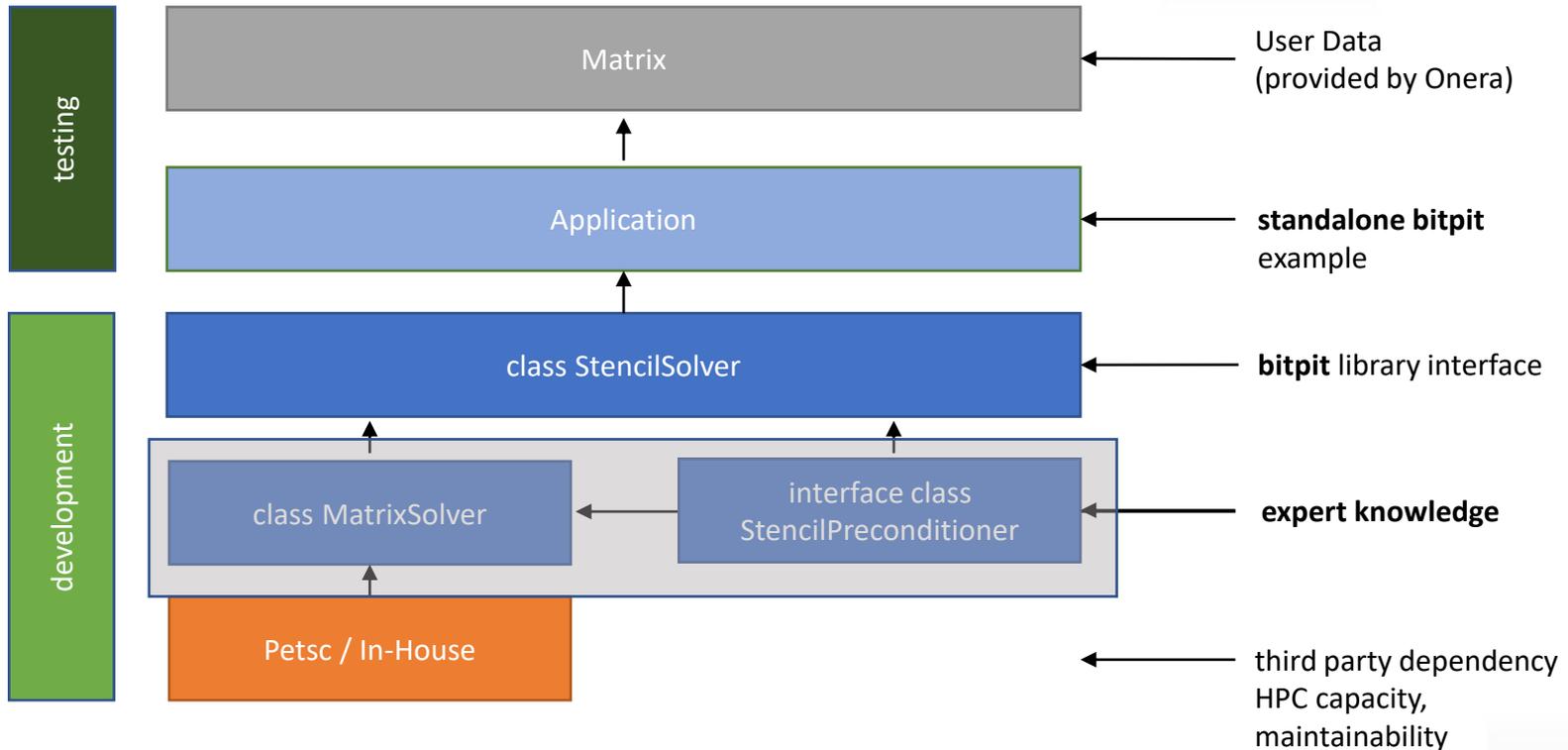


	GMRES(120)		FGMRES(30, 60)	
deflation $k$ eig .	—	36	—	9
mixed precision	—	—	yes	yes
iterations	<i>maxit</i>	11701	747	394
total execution (s)	×	2767	4136	2186
speedup	×	—	1.00	1.89
speedup	×	1.00	—	1.27

Deflation effect on FGMRES  
Best gain over all converging methods

# 2 Optimized preconditioners

- Problem-aware preconditioning



# Test case

2D compressible NS equations

Subsonic laminar flow over NACA0012 profile

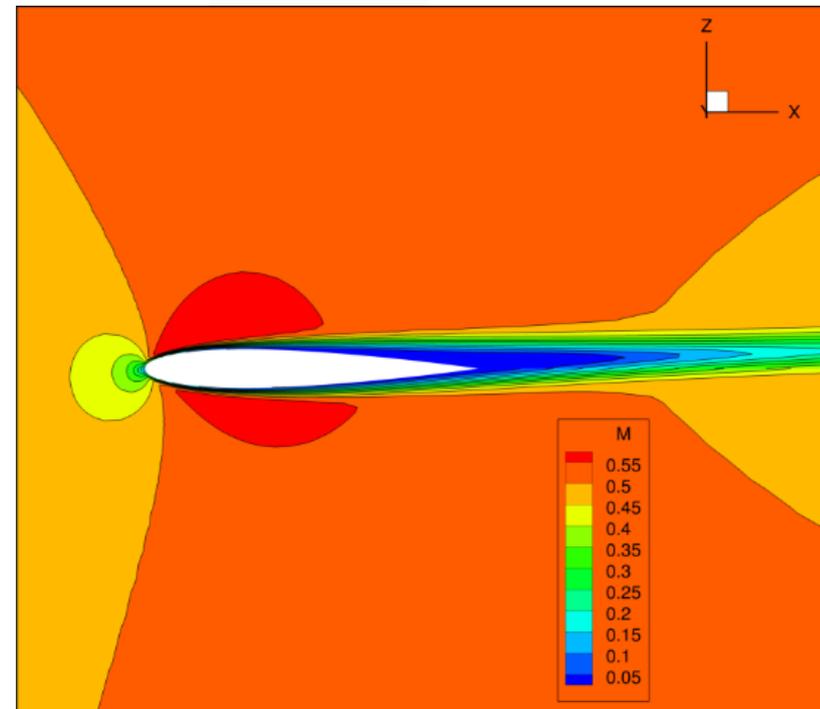
- $M=0.5$ ,  $\alpha=2^\circ$ ,  $Re=5000$

Space discretization

- Mesh is composed of 6400 hexahedra
- 2nd order FV scheme

Aerodynamic shape optimization problem

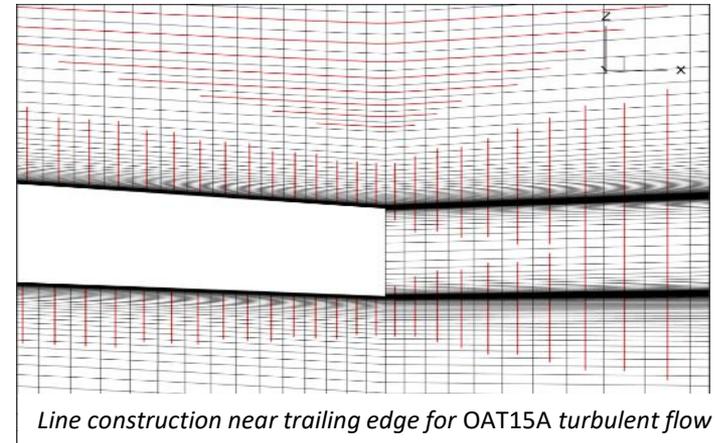
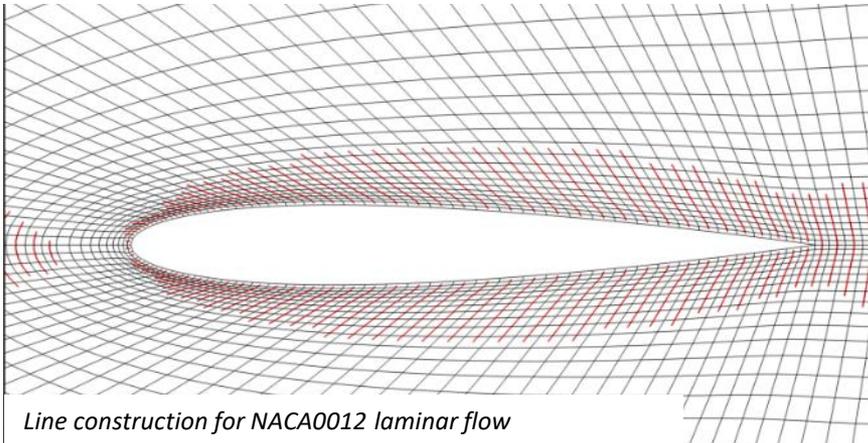
- Minimization of drag coefficient



# Coupling of degrees-of-freedom: creation of lines

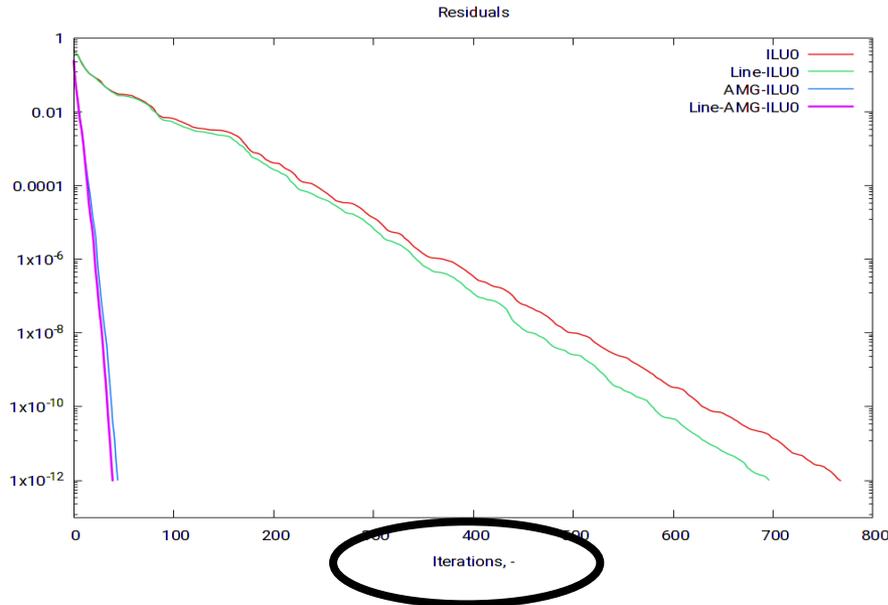
A line construction algorithm to detect strong connections between elements [1]

- Connection's strength is based on the scalar matrix associated to Energy equation
- Each Jacobian entry is associated to a line
- Enlarged inputs for **bitpit** library



[1] An implicit block ILU smoother for preconditioning of Newton-Krylov solvers with application in high-order stabilized finite-element methods, B. R. Ahrabi, D. J. Mavriplis, Computer Methods in Applied Mechanics and Engineering, 358, 2020.

# GMRES + Line-based preconditioner



	Setup time	Solution time	Iterations	Memory	Speedup
<b>ILU0</b>	0.9s	64.6s	769	x1	x 1
<b>Line ILU0</b>	0.9s	53.6s	697	x1	x 1.2
<b>Line AMG</b>	3.9s	26.4s	40	x1.8	x 2.4

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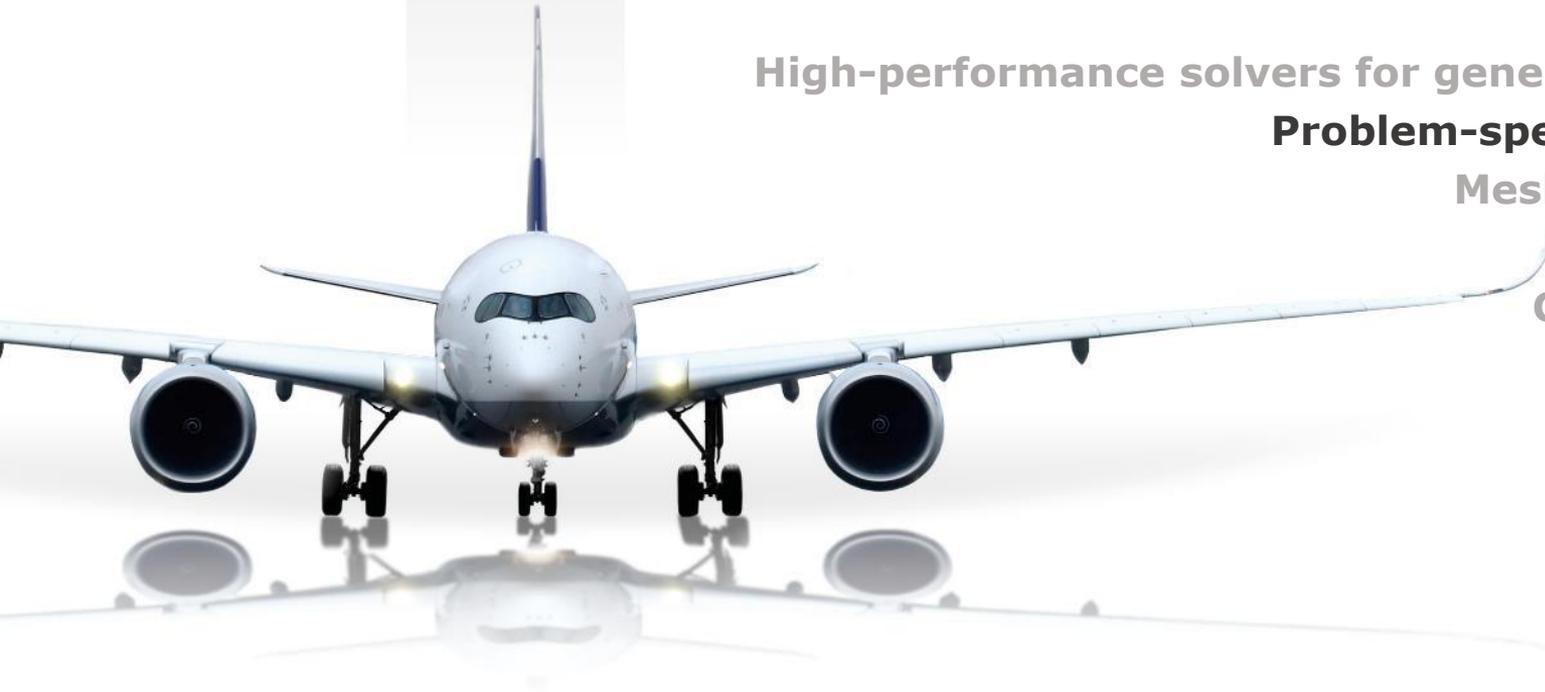
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# Optimize adjoint for periodic flows with phase lags

For turbomachinery (and propellers)

## Zero-phase periodicity

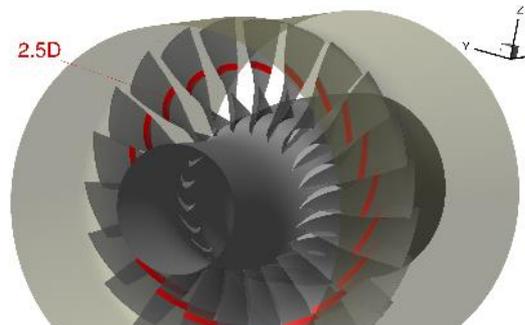
the same phenomena occurs at each blade at each instance

- use periodic boundary conditions on single blade passage

## Phase-lagged periodicity (e.g. distorted inflow)

the same phenomena occurs at each blade but with a phase delay in time

- use full unsteady solver on entire stage



2.5D Rotor 67 with 22 blade passages in a periodically unsteady inflow

# Optimize adjoint for periodic flows with phase lags

## For turbomachinery (and propellers)

### Zero-phase periodicity

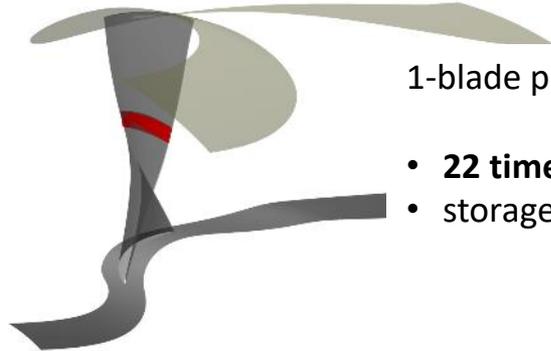
the same phenomena occurs at each blade at each instance

- use periodic boundary conditions on single blade passage

### Phase-lagged periodicity (e.g. distorted inflow)

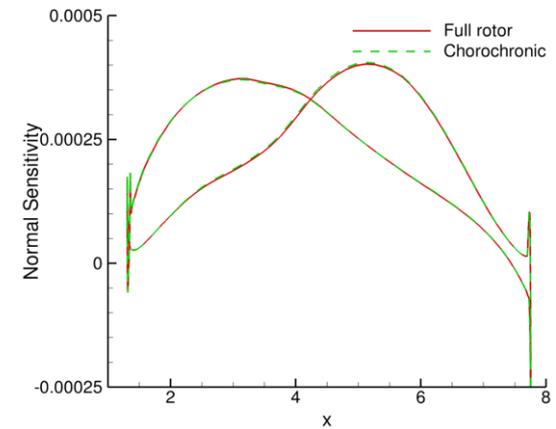
the same phenomena occurs at each blade but with a phase delay in time

- ad-hoc boundary conditions (**chorochronic bc i.e. space-time boundary condition**)
- exploit efficiently this feature; saving ~number of blade



1-blade passage with chorochronic BC

- **22 times reduction of**
- storage, I/O loads and CPU time



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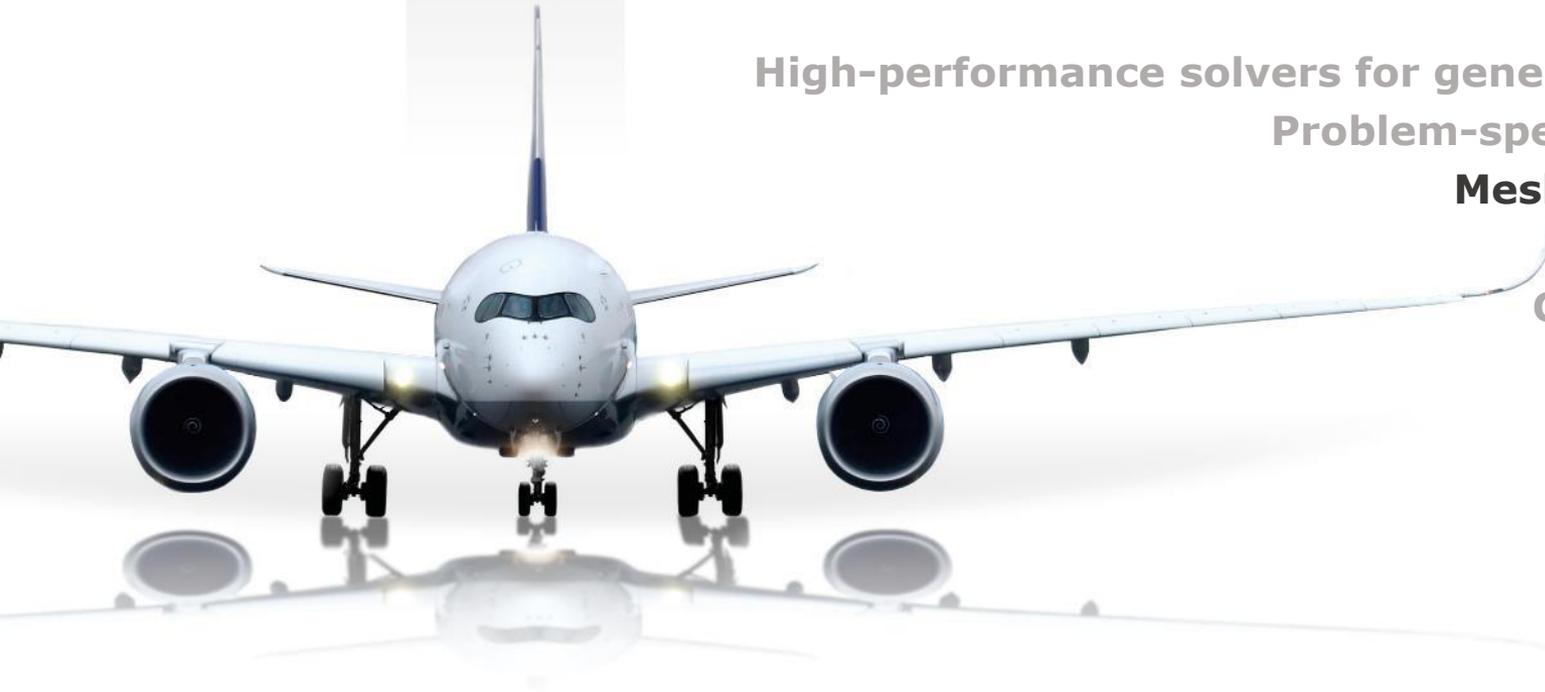
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# Fast mesh morphing

RBF are graph-free interpolation methods

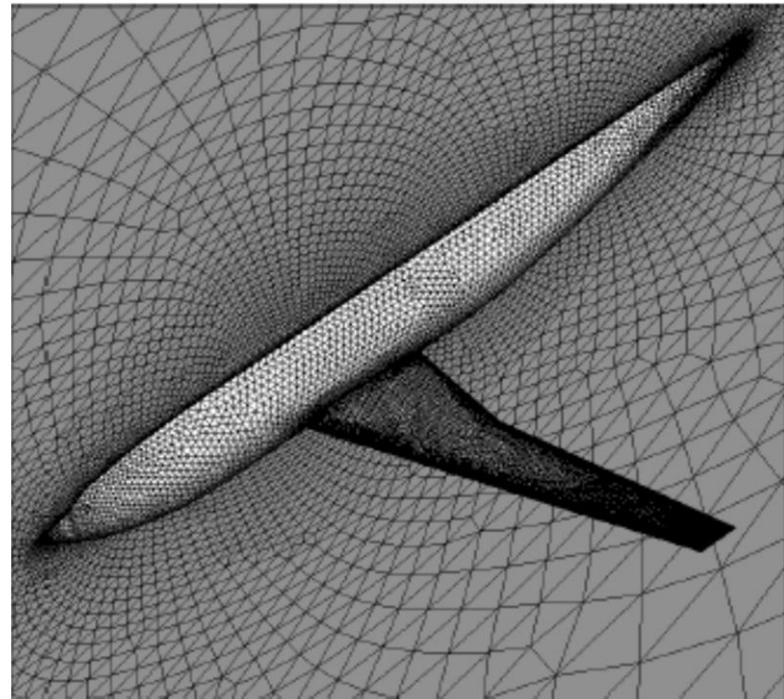
Two steps

## 1 Finding the interpolant

- $N_{\text{surf}} \times N_{\text{surf}}$  solution of a dense linear system
- Industrial case  $N_{\text{surf}} = 10^5$

## 2 Applying the interpolant

- $N_{\text{vol}} \times N_{\text{surf}}$  operations
- Industrial case  $N_{\text{surf}} = 10^5$ ,  $N_{\text{vol}} = 10^{7-8}$
- complexity  $\sim N_{\text{vol}}^{5/3}$   
x10 mesh -> x50 cost  
x100 mesh -> x2500 cost



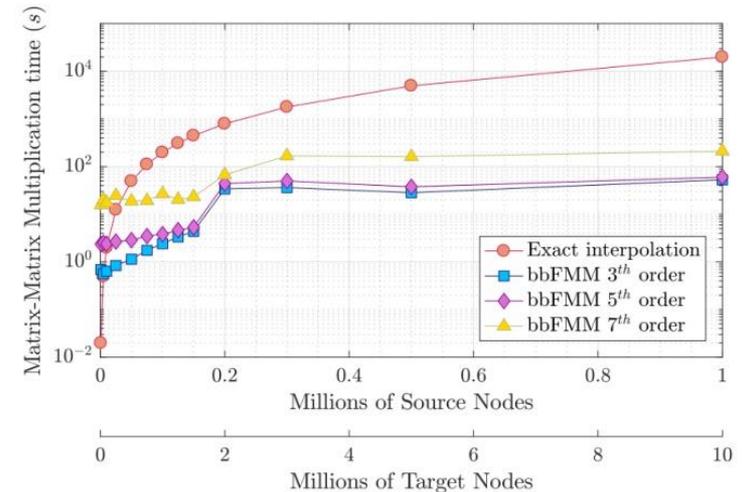
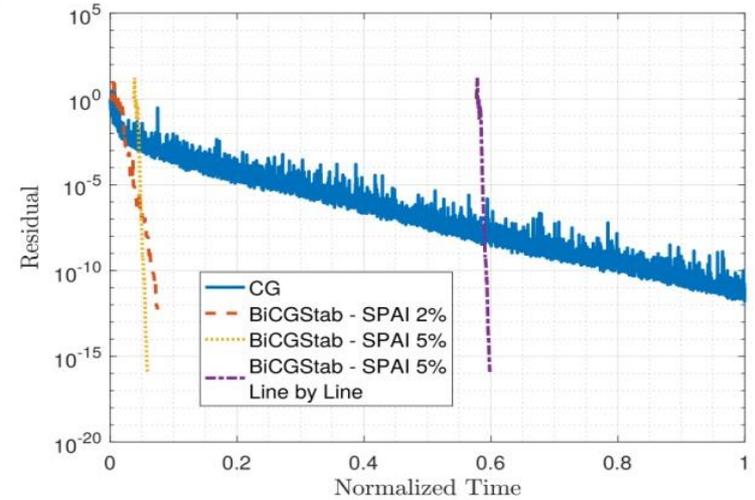
# Fast mesh morphing

## Sparse matrix preconditioner and fast multipole method

- decompose interpolation
  - global sparse contribution
  - local dense correction
- use optimized algorithms for each stage

overall speed-up >10

recover linear complexity



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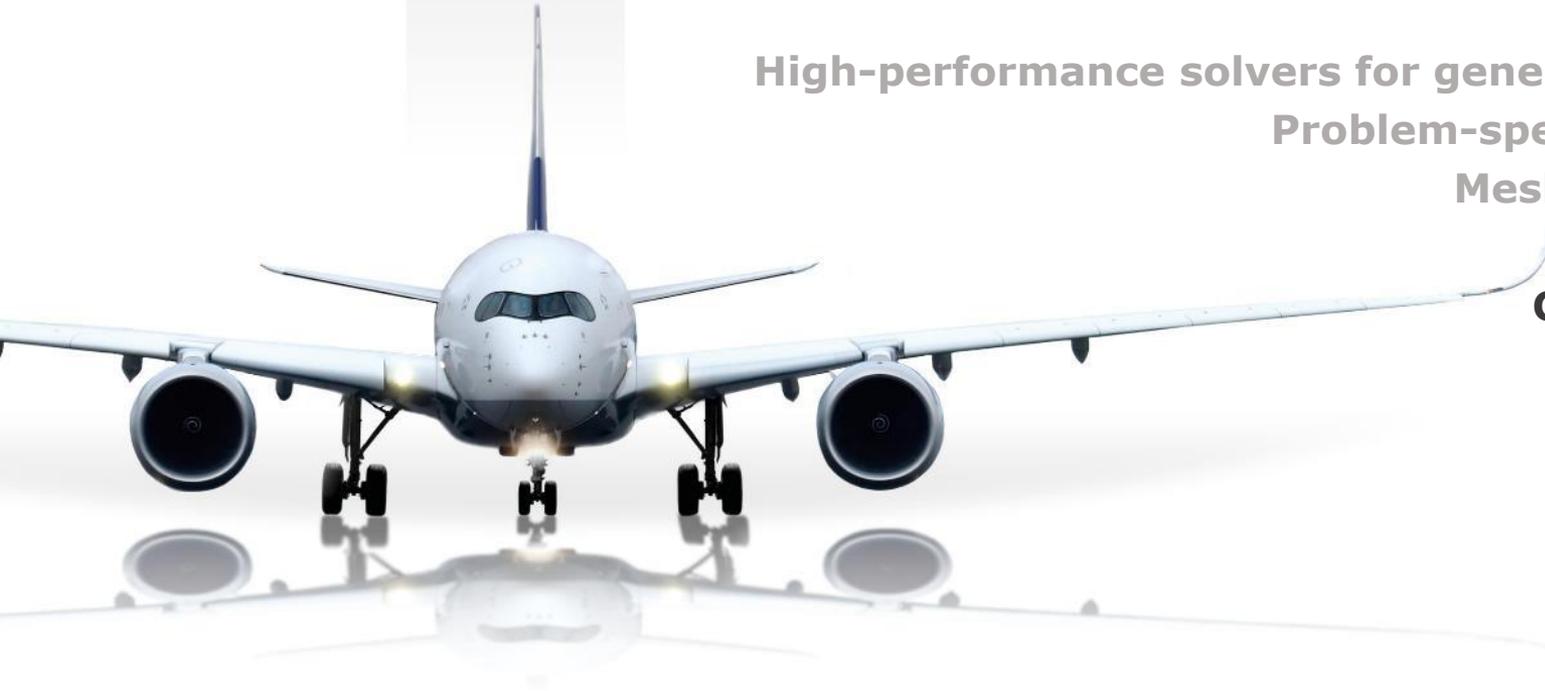
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# Main achievements & lesson learnt

- speed-up for general linear solvers **around 4** (from iterative methods + preconditioners)
- speed-up for unsteady periodic flows with phase-lag **~number of blades**
- speed-up for RBF based mesh morphing **>10**
- **optimization (domain experts/application specific) & maintainability (compliance with next-gen HW) remains an issue**



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